

Airworthiness Directive

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

Amendment 39-4264; **AD 81-06-08**

Airworthiness Directives; Piper Model PA-18-150 Airplanes
PDF Copy (If Available):

▼ Preamble Information

AGENCY: Federal Aviation Administration, DOT

DATES: Effective November 19, 1981.

▼ Regulatory Information

81-06-08 PIPER: Amendment 39-4264. Applies to Piper PA-18-150 Airplanes, S/N 18-8602 thru 18-8109013, certificated in all categories.

Compliance required as indicated.

To prevent failure of the lower fuselage attachment tab which connects the lower horizontal stabilizer brace wire to the fuselage, accomplish the following:

(a) Within the next 10 hours in service after the effective date of this AD, unless already accomplished, conduct a visual inspection of the lower fuselage attachment tab, Piper P/N 80292-15, in the area of the welds for cracks. If cracks are found, install the tail brace wire support in accordance with the Instructions Section of Piper Service Bulletin 706 dated February 23, 1981, or approved alternate method, prior to further flight. If no cracks are found, install the tail brace wire support in accordance with (b) or (c), as appropriate.

(b) For airplanes with less than 500 hours in service on the effective date of this AD, within the next 10 hours in service, unless already accomplished, install the tail brace wire support in accordance with the Instructions Section of Piper Service Bulletin 706 dated February 23, 1981.

(c) For airplanes with 500 hours or more time in service on the effective date of this AD, within the next 100 hours in service, unless already accomplished, install the tail brace wire support in accordance with the Instructions Section of Piper Service Bulletin 706 dated February 23, 1981.

(d) Report findings of cracked parts to the Chief, Engineering and Manufacturing Branch, FAA, Eastern Region, within five days of the inspection; include the time on the part and aircraft serial number. (Reporting approved by the Office of Management and Budget under OMB No. 04-R0174.)

(e) Alternative inspections, alterations, or repairs which provide an equivalent level of safety may be used when approved by the Chief, Engineering and Manufacturing Branch, FAA, Eastern Region.

(f) Upon submission of substantiating data by an owner or operator through an FAA Maintenance Inspector, the Chief, Engineering and Manufacturing Branch, FAA, Eastern Region may adjust the compliance times specified in this AD.

NOTE: Retention of original tab fitting P/N 80292-15 on the airplane is optional.

This amendment is effective November 19, 1981, for all persons except those to whom it was made effective immediately by airmail letter dated March 13, 1981.

▼ Footer Information

▼ Comments