

Airworthiness Directive

▶ Federal Register Information

▼ Header Information

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

AD 63-17-05

Airworthiness Directives; Piper Model PA-25-235 Aircraft

PDF Copy (If Available):

▼ Preamble Information

AGENCY: Federal Aviation Administration, DOT

DATES: Effective September 3, 1963.

▼ Regulatory Information

63-17-05 PIPER: Amdt. 607 Part 507 Federal Register August 22, 1963. Applies to Model PA-25-235 Aircraft Serial Numbers 25-2000 through 25-2519 (Restricted and Normal Category).

Compliance required as indicated.

To preclude weld separations on the end of the landing gear shock strut assembly, Piper P/N 486604, which attaches to the landing gear vee assembly, accomplish the following:

(a) Perform a daily inspection of the specified weld joints for cracks using dye penetrant

or FAA approved equivalent methods until (b) is accomplished. Replace cracked landing gear shock struts before further flight.

(b) Within 25 hours' time in service after the effective date of this AD, unless already accomplished, replace each landing gear shock strut assembly, Piper P/N 486604, with a new shock strut assembly having the same part number but identifiable by the coating of red oxide primer paint.

(Piper Service Bulletin No. 217, dated July 11, 1963, covers this same subject.)

This directive effective September 3, 1963.

▼ Footer Information

▼ Comments

FAA.gov Home | Privacy Policy | Web Policies & Notices | Contact Us | Help

Readers & Viewers: [PDF Reader](#) | [MS Word Viewer](#) | [MS PowerPoint Viewer](#) | [MS Excel Viewer](#) | [WinZip](#)