

Airworthiness Directive

▶ Federal Register Information

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

AD 47-50-12

Airworthiness Directives; Stinson Model 108 Series

PDF Copy (If Available):

▼ Preamble Information

AGENCY: Federal Aviation Administration, DOT

▼ Regulatory Information

47-50-12 STINSON: Applies to Model 108 Series Serial Numbers 1 through 3500.

Compliance required every 100 hours of operation.

Inspection of the stabilizer leading edge attachment to the fuselage should be made for fatigue cracks, after each 100 hours of operation. If fatigue cracks are present, reinforcements to the stabilizer fitting should be added. Inspection may be discontinued after reinforcement is installed.

(Stinson Service Bulletin No. 254 dated September 5, 1947, covers this same subject.)

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Updated RGL applicability to match AD applicability; CAR C-11-185

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