



# SERVICE No. 207A BULLETIN

Piper Aircraft Corporation

Lock Haven, Pennsylvania, U.S.A.

F.A.A. Approved

May 10, 1962

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(Supersedes S. B. No. 207)

**SUBJECT:** Inspection of Weld at Rudder Pedal Assembly  
(Part Nos. 40842-04 and 40842-05)

**MODELS AFFECTED:** PA-18 and PA-18A Super Cub - Serial Nos. 18-1 thru 18-7768  
inclusive

**COMPLIANCE DATE:** Prior to but not later than next 25 hours of flight operation

Field service reports indicate that there have been a few instances of weld separation where the rudder pedal cable attachment fitting is welded to the rudder pedal.

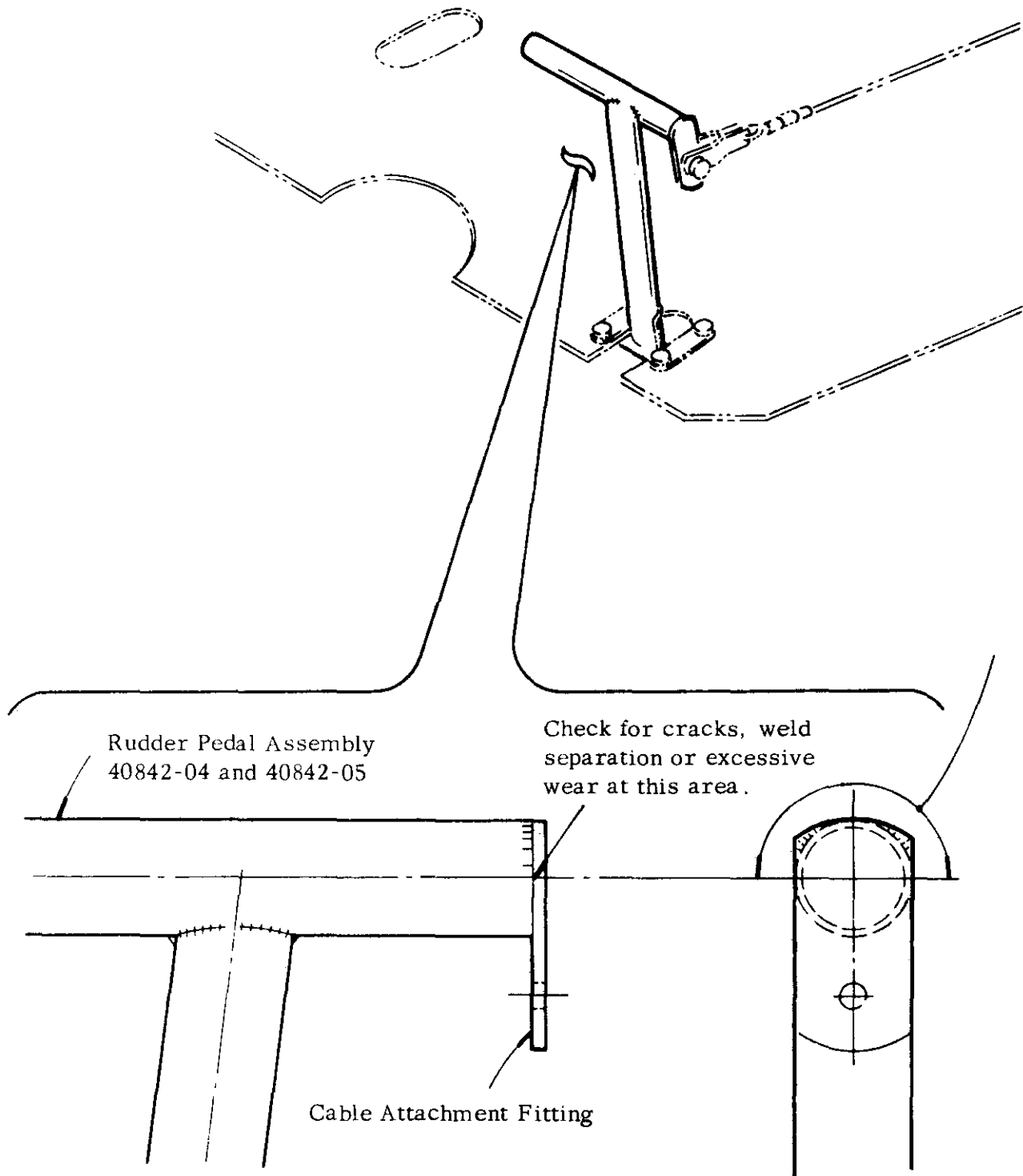
It is requested that a visual inspection be made of this weld for any evidence of separation, cracks in the weld, or excessive wear due to long time usage, in area referred to on the reverse side of this Service Bulletin.

If there is no evidence of weld separation, cracks or wear, the length of the weld should be inspected and if the length of the weld conforms to the dimensions shown in the sketch on the reverse side of this Service Bulletin, no further action is required.

If there is evidence of weld separation, cracks, excessive wear, or the length of the weld does not conform to the dimensions referred to on the sketch; the rudder pedal must be replaced prior to further flight.

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## NOTE

Any cracks in weld, regardless of length of weld or location of crack, shall be cause for rejection of rudder pedal assembly.